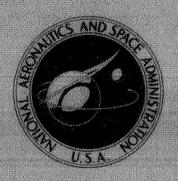
NASA TECHNICAL MEMORANDUM



NASA TM X-2270

CASE FILE

NUMERICAL ANALYSIS OF A PAIR
OF SINGULAR INTEGRAL EQUATIONS
WITH A PRINCIPAL VALUE

by Raymond C. Clerkin Lewis Research Center Cleveland, Ohio 44135

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION - WASHINGTON, D. C. - APRIL 1971

1. Report No. NASA TM X-2270	2. Government Access	on No.	3. Recipient's Catalog	No.	
4. Title and Subtitle			5. Report Date		
NUMERICAL ANALYSIS OF A PAIR OF SINGULAR INTEGRAL			April 1971		
EQUATIONS WITH A PRINCIPAL VALUE			6. Performing Organiza	ation Code	
7. Author(s)			8. Performing Organization Report No.		
Raymond C. Clerkin			E-5974		
Performing Organization Name and Address		10. Work Unit No. 129-04			
Lewis Research Center	 -	11. Contract or Grant	No.		
National Aeronautics and Space Administration			ii. Contract or Grant	ivo.	
Cleveland, Ohio 44135		 -	13. Type of Report and	d Period Covered	
12. Sponsoring Agency Name and Address			Technical Memorandum		
National Aeronautics and Space	 -	14. Sponsoring Agency	Code		
Washington, D.C. 20546			74. Opensoring rigorier	0000	
15. Supplementary Notes					
16. Abstract					
			6		
This report describes a numer			-		
the Volterra type and has a sin	- •		•	-	
of the Fredholm type with a pri		=	-		
value integral is described and	-				
l e e e e e e e e e e e e e e e e e e e	convergence of the iterative solution by the method of Picard it was necessary to take two actions				
to overcome instability near the singularity: (1) reduce the relaxation parameter, and (2) con-					
strain one function to behave smoothly near the singularity.					
				!	
		•			
17. Key Words (Suggested by Author(s))		18. Distribution Statement			
Principal value		Unclassified - unlimited			
Singular integral equations					
Numerical analysis					
19. Security Classif. (of this report)	20. Security Classif. (of this page)		21. No. of Pages	22. Price*	
Unclassified	Unclassified		15	\$3.00	

NUMERICAL ANALYSIS OF A PAIR OF SINGULAR INTEGRAL EQUATIONS WITH A PRINCIPAL VALUE

by Raymond C. Clerkin Lewis Research Center

SUMMARY

This report describes a numerical analysis of a pair of integral equations. One equation is of the Volterra type and has a singularity which requires special handling. The other equation is of the Fredholm type with a principal value.

The technique of integration of the principal value integral is described and tested on an improper integral with a known solution. To attain convergence of the iterative solution by the method of Picard it was necessary to take two actions to overcome instability near the singularity: (1) reduce the relaxation parameter, and (2) constrain one function to behave smoothly near the singularity.

INTRODUCTION

Research on the shape of a magnetically balanced arc reported in reference 1 required the solution of a pair of integral equations. Work done on the numerical solution of this problem has two features of interest to numerical analysts with similar problems: (1) the technique of handling the principal value of an integral, and (2) the method of achieving a satisfactory convergence of the iterative solution of the simultaneous integral equations despite the presence of a singularity.

Because numerical analysts occasionally have need to solve problems involving principal values of integrals or to deal with integral equations having singular points, the experience described in this report should prove helpful.

SYMBOLS

```
a_{1}, b_{1}, c_{1}
              coefficients in definition of f_1, eq. (20)
a_{2}, b_{2}, c_{2}
              coefficients in definition of f_2, eq. (21)
D(s)
              difference in old and new \theta functions, eq. (5b)
f_1, f_2
              quadratic approximations used in eq. (19)
g(s)
              known function of s used as test of accuracy of numerical method used to
                solve for \theta(s)
I_1
              integral to left of principal value integral
Ι,
              principal value integral
              integral to right of principal value integral
I_3
J(s)
              approximating function to numerator integral in eq. (3)
              index number
j
              maximum index for S;
j<sub>max</sub>
K
              refers to the iteration number
              number of points in interval 0 \le s \le 1.0 such that \Delta s = 1/(N - 1)
N
P.V.
              principal value
q
              dummy variable of integration
              integral sum symmetric about q = s (eq. (16))
\mathbf{s}_{\mathbf{j}}
S
              parameter
T_{i}
              \tau function j steps to left of s = 1
              function \theta on K^{th} iteration if there were no relaxation
\mathbf{U}_{\mathbf{K}}
              dummy variable of integration in eq. (19)
X
              limits of integration in eq. (19)
\alpha_1, \alpha_2
              relaxation parameter
β
              variable approaching zero in defining \theta(s) as principal value of integral
\epsilon
\theta(\mathbf{s})
              one of two unknown functions of s
              function \theta(s) at K^{th} iteration
\theta_{\mathbf{K}}(\mathbf{s})
\theta_{0}(s)
              initial guess at function \theta(s)
\tau(s)
              one of two unknown functions of s
```

- $au^*(q)$ function au obtained by expanding $au_K(q)$ in first three terms of its Taylor series about q=s
- $\tau^{\dagger}(s)$ derivative of τ with respect to s
- $\tau^{"}(s)$ second derivative of τ with respect to s
- $\tau_{a}(q)$ function $\tau(q)$ obtained by replacing τ^{*} by its finite difference analog
- $\tau_{K}(s)$ function $\tau(s)$ on K^{th} iteration
- $\tau_{K}^{t}(s)$ derivative of τ with respect to s on K^{th} iteration
- $\tau_{\mathbf{K}}^{tt}(\mathbf{s})$ second derivative of τ with respect to \mathbf{s} on \mathbf{K}^{th} iteration

STATEMENT OF THE PROBLEM

The equations to be solved are

$$\left(\frac{1-\sqrt{1-s^2}}{s}\right)^3 e^{3\tau(s)} = \frac{\int_0^s \sin\left[\theta(q)\right] q \, dq}{\int_0^1 \sin\left[\theta(q)\right] q \, dq} \tag{1}$$

and

$$\theta(s) = P. V. \left[\frac{1}{\pi} \int_{-1}^{1} \frac{\tau(q)}{q - s} dq \right]$$
 (2)

where P.V. is the principal value, and the right side of equation (2) is defined as

$$\lim_{\epsilon \to 0} \left[\frac{1}{\pi} \int_{-1}^{S-\epsilon} \frac{\tau(q)dq}{q-s} + \frac{1}{\pi} \int_{S+\epsilon}^{1} \frac{\tau(q)dq}{q-s} \right]$$

The quantities $\tau(s)$ and $\theta(s)$ are the unknown functions over the interval $0 \le s \le 1$. $\tau(-s) = \tau(s)$ defines τ in the interval $-1 \le s \le 0$. Equation (1) is nonlinear and of the

Volterra type (ref. 2). If taken as an equation for $\tau(s)$, it has a singularity at s=0.

Equation (2) is linear and of the Fredholm type. It has a singularity at q = s.

The solution of the problem depends on the proper handling of these two singularities. The given conditions on $\tau(s)$ and $\theta(s)$ are

$$\tau(s) = \tau(-s)$$
 with $\tau(1) = 0$

$$\theta(s) = -\theta(-s)$$
 with $\theta(0) = 0$

where $\tau(s)$ is an even function of s, and $\theta(s)$ is an odd function of s.

METHOD OF SOLUTION

The numerical solution is obtained by using Picard's method of iteration. Equations (1) and (2) are rewritten as

$$e^{3\tau_{K}(s)} = \left(\frac{s}{1 - \sqrt{1 - s^{2}}}\right)^{3} \frac{\int_{0}^{s} q \sin\left[\theta_{K-1}(q)\right] dq}{\int_{0}^{1} q \sin\left[\theta_{K-1}(q)\right] dq}$$
(3)

$$U_{K}(s) = P.V. \left[\frac{1}{\pi} \int_{-1}^{1} \frac{\tau_{K}(q)dq}{q-s} \right]$$
(4)

The next estimate $\theta_{\mathbf{K}}$ is defined by

$$\theta_{K}(s) = \theta_{K-1}(s) + \beta D(s)$$
 (5a)

where D is the suggested correction to θ_{K-1} defined by

$$D(s) = U_{K}(s) - \theta_{K-1}(s)$$
 (5b)

and β is the relaxation parameter.

The three steps of the iterative loop are briefly

- (1) Solve equation (3) for $\tau_{K}(s)$.
- (2) Solve equation (4) for $U_K(s)$.
- (3) Combine θ_{K-1} and U_K by equations (5a) and (5b) to get a new estimate θ_K . The starting estimate used was

$$\theta_{O}(s) = \frac{-\pi s}{4} \tag{6}$$

Two aspects of the iterative process will be discussed in greater detail in the following sections:

- (1) In the section INTEGRATION TECHNIQUE, the principal value
- (2) In the section ACHIEVEMENT OF CONVERGENCE, convergence of the iterative process by
 - (a) Proper handling of the singularity in equation (3)
 - (b) Proper choice of the relaxation parameter β of equation (5a)

INTEGRATION TECHNIQUE

To evaluate U_K from equation (4) requires a technique for dealing with the principal value of an integral. Since the singularity is at q = s, and τ_K is known at all s points, -1, $-1 + \Delta s$, ..., $1 - \Delta s$, and 1, the integral is split up into at most three parts, with the second, I_2 , containing the singularity:

$$U_{K} = I_{1} + I_{2} + I_{3}$$

$$= \int_{-1}^{S - \Delta S} + \int_{S - \Delta S}^{S + \Delta S} + \int_{S + \Delta S}^{1.00}$$
(7)

The I_1 and I_3 integrals may be evaluated by Simpson's rule. The I_2 integral is treated as follows:

For $s=0, \Delta s, 2\Delta s, \ldots$, and $1-\Delta s$ expand τ_K in a Taylor series around q=s and truncate after three terms to get the approximation τ^* :

$$\tau^*(q) = \tau_{K}(s) + \tau_{K}^{\dagger}(s)(q - s) + \tau_{K}^{\dagger\dagger}(s) \frac{(q - s)^2}{2}$$
 (8)

Replace equation (8) by its finite difference analog to get the approximation $\tau_{\rm a}({\bf q})$:

$$\tau_{\mathbf{a}}(\mathbf{q}) = \tau_{\mathbf{K}}(\mathbf{s}) + \frac{\tau_{\mathbf{K}}(\mathbf{s} + \Delta \mathbf{s}) - \tau_{\mathbf{K}}(\mathbf{s} - \Delta \mathbf{s})}{2\Delta \mathbf{s}} (\mathbf{q} - \mathbf{s})$$

$$+ \frac{\tau_{K}(s + \Delta s) - 2\tau_{K}(s) + \tau_{K}(s - \Delta s)}{\Delta s^{2}} \frac{(q - s)^{2}}{2}$$
(9)

Replace $\tau_{K}(q)$ in I_{2} by $\tau_{a}(q)$ to get

$$I_{2}(s) = \frac{\tau_{K}(s)}{\pi} \int_{s-\Delta s}^{s+\Delta s} \frac{dq}{q-s} + \frac{\tau_{K}(s+\Delta s) - \tau_{K}(s-\Delta s)}{2\Delta s\pi} \int_{s-\Delta s}^{s+\Delta s} dq$$

$$+\frac{\tau_{K}(s + \Delta s) - 2\tau_{K}(s) + \tau_{K}(s - \Delta s)}{\pi \Delta s^{2}} \int_{s - \Delta s}^{s + \Delta s} \frac{q - s}{2} dq \qquad (10)$$

The principal value of the first integral is zero. The second integral is $2\Delta s$, and the third integral vanishes to yield

$$I_2(s) = \frac{\tau_K(s + \Delta s) - \tau_K(s - \Delta s)}{\pi}$$
 for $s = 0, \Delta s, ..., 1 - \Delta s$ (11)

At s = 1, I_2 reduces to

$$\int_{1-\Delta S}^{1} \frac{\tau_{K}(q)dq}{q-1}$$

For this point, all derivatives appearing in the Taylor series expansion of $\tau_K(q)$ about the point q=1 are one-sided. Use T_j to denote $\tau_K(1-j\Delta s)$; then, since $T_0=0$, the equations for the finite difference analogs of τ^i and τ^{ij} become

$$\tau_{K}^{*}(1) = \frac{-T_{1}}{\Delta s} + 0.5 \left(-\frac{T_{1}}{\Delta s} - \frac{T_{1} - T_{2}}{\Delta s} \right) = \frac{T_{2} - 4T_{1}}{2\Delta s}$$
(12)

and

$$\tau_{K}^{"}(1) = \frac{-2T_{1} + T_{2}}{\Delta s^{2}} + \left(\frac{-2T_{1} + T_{2}}{\Delta s^{2}} - \frac{T_{1} - 2T_{2} + T_{3}}{\Delta s^{2}}\right)$$

$$= \frac{-5T_{1} + 4T_{2} - T_{3}}{\Delta s^{2}}$$
(13)

Substitute equations (12) and (13) in equation (8) for s=1 to get the approximation $\tau_{a}(q)$:

$$\tau_{a}(q) = \frac{T_{2} - 4T_{1}}{2\Delta s} (q - 1) + \frac{-5T_{1} + 4T_{2} - T_{3}}{\Delta s^{2}} \frac{(q - 1)^{2}}{2}$$
(14)

This formula is used in the equation for

$$I_2(1) = \frac{1}{\pi} \int_{1-\Delta s}^{1} \frac{\tau_a(q)dq}{q-1}$$
 (15a)

to yield

$$I_{2}(1) = \frac{-3T_{1} - 2T_{2} + T_{3}}{4\pi}$$

$$= \frac{-3\tau_{K}(1 - \Delta s) - 2\tau_{K}(1 - 2\Delta s) + \tau_{K}(1 - 3\Delta s)}{4\pi}$$
(15b)

The evaluation of I_1 and I_3 was handled first by Simpson's rule formulas. Later a second method was applied. Both methods yielded sufficient accuracy for this problem and for the test case. But other problems involving principal value integration may be nefit from the more careful technique of the second method. Therefore, its derivation is given here.

Note that the I_1 integrand $\tau_K(q)/(q-s)$ is large and of one sign near $s-\Delta s$, whereas the I_3 integrand is large and of opposite sign near $s+\Delta s$. Using differences of large numbers may increase the truncation error. Combining symmetric portions of

 I_1 and I_3 should reduce the truncation error. Let S_i be the integral sum

$$S_{j} = \int_{s-(j+1)\Delta s}^{s-j\Delta s} \frac{\tau_{K}(q)dq}{q-s} + \int_{s+j\Delta s}^{s+(j+1)\Delta s} \frac{\tau_{K}(q)dq}{q-s}$$
(16)

where $j = 1, 2, 3, \ldots, j_{max}$ and

$$j_{\text{max}} = N - i \tag{17}$$

where i is such that $s=(i-1)\Delta s$ and N is the number of points in $0 \le s \le 1$ such that $\Delta s=1/(N-1)$. Then the sum of I_1 and I_3 becomes

$$I_{1}(s) + I_{3}(s) = \int_{-1}^{2s-1} \frac{\tau_{K}(q)dq}{q-s} + \sum_{j=1}^{j_{\max}} S_{j}$$
 (18)

where much of the important cancellation of large numbers occurs in the computing of S_i for low j.

The S_j are computed as follows. Make use of a change of variables to obtain for equation (16)

$$S_{j} = \int_{-\alpha_{2}}^{-\alpha_{1}} \frac{f_{1}(x)dx}{x} + \int_{\alpha_{1}}^{\alpha_{2}} \frac{f_{2}(x)dx}{x}$$
 (19)

where $\alpha_1 < \alpha_2$.

Fit a parabola through the three values of f_1 at $x = -\alpha_2$, $-\alpha_1$, and $\alpha_2 - 2\alpha_1$ to get

$$f_1(x) = a_1 + b_1 x + c_1 x^2 \tag{20}$$

and fit a parabola through the three values of f_2 at $x = 2\alpha_1 - \alpha_2$, α_1 , and α_2 to get

$$f_2(x) = a_2 + b_2 x + c_2 x^2$$
 (21)

Then S_i may be integrated in closed form to get

$$S_{j} = a_{1} \ln \frac{\alpha_{1}}{\alpha_{2}} + b_{1}(\alpha_{2} - \alpha_{1}) + \frac{c_{1}}{2} (\alpha_{1}^{2} - \alpha_{2}^{2})$$

$$+ a_2 \ln \frac{\alpha_2}{\alpha_1} + b_2(\alpha_2 - \alpha_1) + \frac{c_2}{2} \left(\alpha_2^2 - \alpha_1^2\right)$$
 (22)

or

$$S_{j} = (a_{2} - a_{1}) \ln \frac{\alpha_{2}}{\alpha_{1}} + (b_{1} + b_{2})(\alpha_{2} - \alpha_{1}) + \frac{(c_{2} - c_{1})}{2} (\alpha_{2}^{2} - \alpha_{1}^{2})$$
 (23)

As a test, this integration process was applied to the function

$$g(s) = \int_{-1}^{1} \frac{(1 - q^2)dq}{q - s}$$
 (24)

The numerator of the integrand in equation (24) is like τ in being even and zero at q = 1. The exact value of g(s) is

$$g(s) = (1 - s2) ln \frac{1 - s}{1 + s} - 2s$$
 (25)

Note that g(1) = -2.

The following table gives the integral of equation (24) for values of $\, s \,$ together with the correct value of $\, g(s)$. The integration technique is clearly accurate.

Distance	Integral of eq. (24)		
parameter, s	Numerical value	Correct value	
0.1	-0.39866259	-0.39866398	
. 2	78924440	78924650	
. 3	-1.1633240	-1.1633257	
. 4	-1.5117280	-1.5117302	
. 5	-1.8239566	-1.8239592	
. 6	-2.0872253	-2.0872283	
.7	-2.2846440	-2.2846465	
.8	-2.3909990	-2.3910008	
.9	-2.3594423	-2.3594434	
. 99	-2.0853368	-2.0853368	

ACHIEVEMENT OF CONVERGENCE

One action necessary to achieve convergence is to handle properly the singularity in the equation for τ at s = 0. In equation (3), the term in brackets can be written as

$$\frac{s}{1 - \sqrt{1 - s^2}} \sim \frac{s}{1 - \left(1 - \frac{1}{2}s^2 - \frac{1}{8}s^4\right)} = \frac{2}{s\left(1 + \frac{s^2}{4}\right)}$$
(26)

Its cube behaves like s^{-3} near s = 0.

Since the upper integral on the right side of equation (3) behaves like

$$J(s) = \int_0^s q \sin(Kq) dq \sim \int_0^s Kq^2 dq = \frac{Ks^3}{3}$$
 (27)

it becomes obvious that the right side of equation (3) behaves like s^3/s^3 in the vicinity of zero.

Because of the singularity at zero, the numeric difficulties become more severe as s approaches zero. Any errors generated in the evaluation of the upper integral are magnified when divided by s^3 and lead to instability. Since τ is known to be an even function of s with zero derivative at s=0, it seems logical to impose proper behavior in the neighborhood of zero numerically.

Another necessary action to achieve convergence was to choose the proper relaxation parameter β . (See eq. (5a) for its definition.)

The use of $\beta = 1.0$ as the coefficient of the suggested change led to an oscillation of about 5 percent in the value of $\tau(0)$ (see fig. 1). Figure 2 shows how the oscillation drops to less than 3 percent when β is 0.9. Figure 3 shows how the oscillation drops to less than 0.5 percent when β is 0.7. Figure 4 shows that with $\beta = 0.5$ the oscillation disappears, and convergence is very good after 18 iterations.

The converged shape of the $\tau(s)$ function is shown in figure 5. The converged shape of $\theta(s)$ is shown in figure 6.

CONCLUDING REMARKS

This report shows that there were two necessary steps required to achieve the solution of the original system of two integral equations. The steps taken were

- (1) The $\tau_{\mathbf{K}}(\mathbf{s})$ function was isolated in the neighborhood of $\mathbf{s}=\mathbf{0}$, and a smoothing process was applied to remove the erratic behavior of $\tau_{\mathbf{K}}$ near $\mathbf{s}=\mathbf{0}$.
- (2) A relaxation factor equal to about 0.5 was used to remove the oscillations of the $\tau_{\rm K}({\rm s})$ curve.

This method, then, provided an accurate solution to the system of two integral equations, as is shown in the section of the report on accuracy.

Lewis Research Center,

National Aeronautics and Space Administration, Cleveland, Ohio, February 10, 1971, 129-04.

REFERENCES

- 1. Goldstein, Marvin E.; and Braun, Willis H.: Shape of a Magnetically Balanced Arc. NASA TN D-4736, 1968.
- 2. Hildebrand, Francis B.: Methods of Applied Mathematics. Prentice-Hall, Inc., 1952.

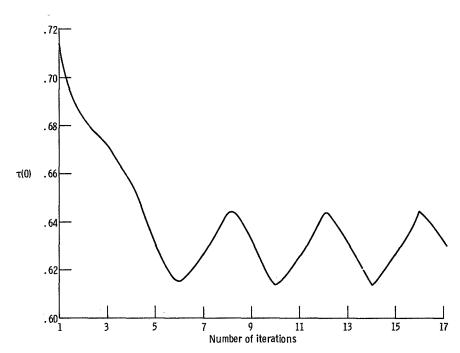


Figure 1. - Behavior of $\tau(0)$ value with successive iterations with relaxation parameter β = 1.

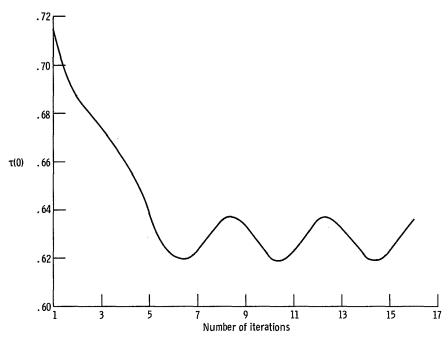


Figure 2. - Behavior of $\tau(0)$ value with successive iterations with relaxation parameter β = 0.9.

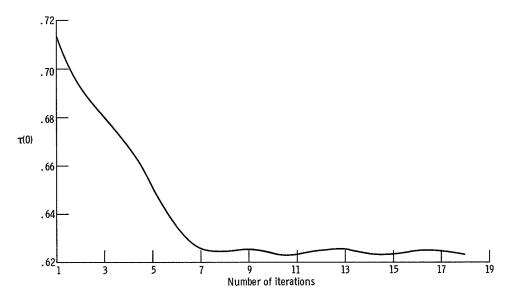


Figure 3. - Behavior of $\tau(0)$ value with successive iterations with relaxation parameter β = 0.7.

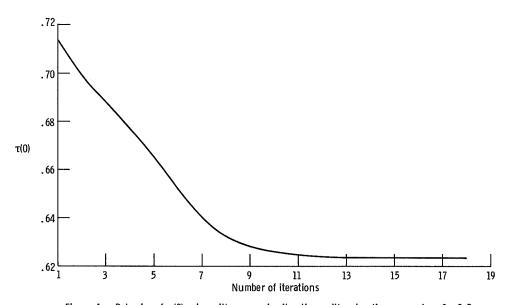


Figure 4. – Behavior of $\tau(0)$ value with successive iterations with relaxation parameter β = 0.5.

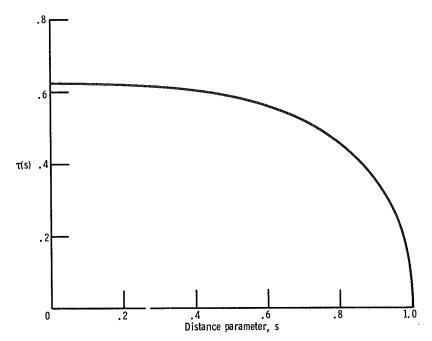


Figure 5. - Plot of converged $\,\tau\,$ as function of distance parameter $\,$ s.

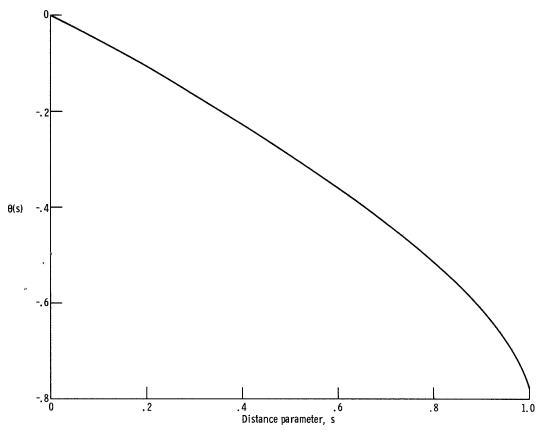


Figure 6. - Plot of converged $\,\theta\,$ as function of distance parameter $\,$ s.

NATIONAL AFRONAUTICS AND SPACE ADMINISTRATION WASHINGTON, D. C. 20546

OFFICIAL BUSINESS
PENALTY FOR PRIVATE USE \$300

FIRST CLASS MAIL



POSTAGE AND FEES PAID NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

POSTMASTER:

If Undeliverable (Section 158 Postal Manual) Do Not Return

"The aeronautical and space activities of the United States shall be conducted so as to contribute... to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof."

-NATIONAL AFRONAUTICS AND SPACE ACT OF 1958

NASA SCIENTIFIC AND TECHNICAL PUBLICATIONS

TECHNICAL REPORTS: Scientific and technical information considered important, complete, and a lasting contribution to existing knowledge.

TECHNICAL NOTES: Information less broad in scope but nevertheless of importance as a contribution to existing knowledge.

TECHNICAL MEMORANDUMS: Information receiving limited distribution because of preliminary data, security classification, or other reasons.

CONTRACTOR REPORTS: Scientific and technical information generated under a NASA contract or grant and considered an important contribution to existing knowledge.

TECHNICAL TRANSLATIONS: Information published in a foreign language considered to merit NASA distribution in English.

SPECIAL PUBLICATIONS: Information derived from or of value to NASA activities. Publications include conference proceedings, monographs, data compilations, handbooks, sourcebooks, and special bibliographies.

TECHNOLOGY UTILIZATION
PUBLICATIONS: Information on technology used by NASA that may be of particular interest in commercial and other non-aerospace applications. Publications include Tech Briefs, Technology Utilization Reports and Technology Surveys.

Details on the availability of these publications may be obtained from:

SCIENTIFIC AND TECHNICAL INFORMATION OFFICE

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

Washington, D.C. 20546